

[1]王明昊,刘刚,何志昆,等.基于LFT-LPV结构的导弹自动驾驶仪降保守性设计[J].弹箭与制导学报,2013,2:1-3.

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点击复

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基于LFT-LPV结构的导弹自动驾驶仪降保守性设计

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Title: Reduction of Conservatism in Missile Autopilot Based on LFT-LPV

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关键词: 线性变参数; 线性分式变换; 全块比例矩阵; 输出反馈; 线性矩阵不等式

Keywords: linear parameter system; linear fractional transformation; full block scaling matrix; output feedback; LMI

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摘要: 针对一类导弹纵向运动模型,提出了一种基于全块比例矩阵的LFT-LPV系统降保守性输出反馈控制器的设计方法。首先在感兴趣的飞行区域内通过数据拟合得到系统模型的LFT-LPV表示,利用S-过程,通过选择具有特定结构的全块比例矩阵避免需要在所有可能的变参数轨迹上求解无穷个LMIs的问题,同时推导出满足LFT-LPV系统性能指标的输出反馈控制器的设计方法。最后的仿真结果验证了方法的有效性。

Abstract: According to a class of longitudinal model of missile, an output feedback controller design methodology of LFT-LPV system based on full block scaling matrix is proposed which reduce the conservatism effectively. Firstly, the system nonlinear model is transformed into a LFT-LPV form by data fitting in the interested working region. Then, using S-Procudure, full block scaling matrix with a specific configuration is chosen to satisfy the inequality, avoid solving a infinite numbers of LMIs on the trajectory of all possible varying parameters, and then give the method of the output feedback controller design which can satisfy the given performance target based on LFT-LPV system. Finally, the digital simulation results prove the availability of the proposed method.

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