[1]黄巍,王跃钢,沈琰,等.导弹发射车SINS/GNS定位定向技术[J].弹箭与制导学报,2009,3:59.





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导弹发射车SINS/GNS定位定向技术(PDF)

《弹箭与制导学报》[ISSN:1673-9728/CN:61-1234/TJ] 期数: 2009年第3期 页码: 59 栏目: 导弹与 制导技术 出版日期: 2009-06-25

Title: SINS/GNS Positioning and Orientation for Missile Launching Vehicle

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关键词: 捷联惯性导航系统; 地磁匹配; 定位定向; U卡尔曼滤波; 仿真

SINS; geomagnetic matching; positioning and orientation; UKF; simulation Keywords:

TJ768 2 分类号:

DOI:

文献标识码: A

针对导弹发射车对导航系统的全自主性及精度的特殊要求,提出将地磁匹配定位定向系 摘要:

> 统与惯性导航系统进行组合。地磁匹配过程中采用了基于 Hausdorff 距离的相似算 法。基于U卡尔曼滤波算法,对SINS/GNS全自主定位定向组合系统与捷联惯导系统单独 定位定向在同一条件下的误差情况进行了对比仿真。仿真结果表明,装有地磁匹配惯导

系统的导弹发射车定位定向的精确度有了明显提高,滤波收敛性较好。

Abstract: Focused on the special requirements for independence and precision of

> navigation system in missile launching vehicle, the navigation program which matches inertial guidance system with geomagnetic matching positioning and orientation system was put forward. The similar algorithm based on Hausdorff range was adopted in the geomagnetic matching process. The SINS/GNS independent positioning and orientation system was designed under the same condition, the two different errors (SINS/GNS and SINS) were compared and simulated based on the UKF. The simulation result shows that the positioning and orientation accuracy of the missile launching vehicle equipped with SINS/GNS is improved evidently and the filter convergence has great reliability.

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