## 航空动力学报

中国航空学会主

首页 本刊介绍 编委会 投稿须知 审稿编辑流程 期刊征订 广告征订

English 选择皮肤:

Hide Expanded Menus

张磊, 邱志平. 碳纤维增强点阵夹芯结构的屈曲强度[J]. 航空动力学报, 2013, 28(3):525~530

## 碳纤维增强点阵夹芯结构的屈曲强度

Buckling of lattice truss cores sandwich structures with carbon fiber reinforced

投稿时间: 2012-03-21

DOI:

中文关键词: 碳纤维增强 点阵夹芯结构 屈曲强度分析 位移场假设 有限单元法

英文关键词:carbon fiber reinforced lattice truss cores sandwich structure buckling analysis displacement field assumption finite

element method

基金项目: 国家自然科学基金重大研究计划(90816024); 高等学校学科创新引智计划(B07009)

作者

北京航空航天大学 航空科学与工程学院,北京 100191 张磊 邱志平 北京航空航天大学 航空科学与工程学院,北京 100191

摘要点击次数: 305

全文下载次数: 568

中文摘要:

考虑到点阵芯层的结构形式及杆单元以拉压为主的变形模式, 基于连续介质等效理论, 提出了一种全新的位移场假设, 并以此编写了计算点阵夹芯复合 材料屈曲强度的有限元程序. 通过与不同试样的压缩试验的比较, 对碳纤维增强点阵夹芯结构的屈曲行为进行了分析, 并且验证了程序的有效性. 同时, 采用程 序, 讨论了点阵参数(包括杆单元长度、半径及倾斜角)对点阵夹芯复合材料结构屈曲强度的影响规律. 结果表明: 芯层体积分数相同的情况下, 金字塔型和四面 体型点阵夹芯结构的屈曲强度相差不大,后者略高,所得结论对点阵夹芯复合材料结构设计具有一定的指导意义.

Considering the geometrical configuration of the structure and the deformation mode of the lattice truss, a new displacement field assumption was proposed based on the equivalent continuum theory, and finite element method program was prepared for buckling analysis of lattice truss-core composites. By comparing with the compression test of a group of samples, the buckling behavior of sandwich structures with carbon fiber reinforced lattice truss cores was studied, and the validity was affirmed Furthermore, the relationship between the buckling load of lattice truss-core composites and lattice parameters (including length, radius and angle of lattice truss) was discussed. The results demonstrate that the little difference exists in bulking load between pyramidal and tetrahedral lattice truss sandwich structures when the volume fraction of the cores are the same, the latter is a little higher. The conclusions are meaningful for the design of lattice truss-core composites.

查看全文 查看/发表评论 下载PDF阅读器