



航空学报 » 2012, Vol. 33 » Issue (12) :2211-2220 DOI:

固体力学与飞行器总体设计

[最新目录](#) | [下期目录](#) | [过刊浏览](#) | [高级检索](#)

[<<](#) [<](#) [前一篇](#) | [后一篇](#) [>](#) [>>](#)

疲劳裂纹尖端残余应力场的深度-传感压痕测试与有限元分析

刘建中¹, 叶笃毅², 张丽娜¹, 肖磊²

1. 中国航空工业集团公司 北京航空材料研究院, 北京 100095;
2. 浙江大学 化工机械研究所, 浙江 杭州 310027

Depth-sensing Indentation Measurement and Finite Element Analysis of Residual Stress Field near Fatigue Crack Tip

LIU Jianzhong¹, YE Duyi², ZHANG Li' na¹, XIAO Lei²

1. Beijing Institute of Aeronautical Materials, China Aviation Industry Corporation, Beijing 100095, China;
2. Institute for Process Equipments, Zhejiang University, Hangzhou 310027, China

摘要

参考文献

相关文章

Download: [PDF \(4162KB\)](#) [HTML KB](#) Export: [BibTeX](#) or [EndNote \(RIS\)](#) [Supporting Info](#)

摘要

飞机和发动机等重要装备承力结构在服役过程中通常承受变幅疲劳载荷作用。直接测量和分析由于过载塑性变形而导致的裂纹尖端附近残余应力场,对于较好地理解变幅加载下疲劳裂纹扩展行为,从而改善和发展疲劳寿命预测模型具有重要价值。本文基于微观尺度的深度-传感压痕(DSI)残余应力测量技术,研究了材料疲劳裂纹尖端附近残余应力场的实用测试技术,获得了铝合金中心裂纹拉伸试样在恒幅及单峰疲劳过载作用下裂纹尖端附近的残余应力场分布。同时,还采用弹塑性有限元方法模拟分析了相同疲劳载荷下裂纹尖端附近相应的残余应力场分布。相互验证表明:两种方法获得了基本吻合的结果。

关键词: 疲劳裂纹扩展 单峰过载 裂纹尖端残余应力 弹塑性有限元 深度-传感压痕技术

Abstract:

The aircraft, engine and other key load bearing structures are often subjected to variable-amplitude fatigue loading in the course of their services. Direct measurement and analysis of the residual stress field near a crack tip caused by plastic deformation due to overload is of great value for a better understanding of fatigue crack growth behavior under variable-amplitude loading, and thus for the improvement and development of a fatigue life prediction model. Based on the microscale depth-sensing indentation (DSI) residual stress measurement technique, this paper studies the practical testing of the residual stress field near a fatigue crack tip, obtains the distribution of residual stress field near it in fatigue crack propagation process for a middle-crack tensile specimen of an aluminum alloy under constant amplitude and single peak overload. Also, it analyzes the residual stress distribution near the crack tip using the elastic-plastic finite element method under the same fatigue loads. Mutual authentication indicates that the results obtained by the two different ways agree well with each other.

Keywords: fatigue crack propagation single peak overload residual stress near crack tip elastic-plastic finite element depth-sensing indentation technique

Received 2012-02-17;

Corresponding Authors: 刘建中 Email: jianzhongliu09@sina.com

About author: 刘建中 男, 博士, 研究员, 博士生导师。主要研究方向: 材料与结构力学行为, 疲劳与断裂力学、可靠性。Tel: 010-62496701 E-mail: jianzhongliu09@sina.com

引用本文:

刘建中, 叶笃毅, 张丽娜, 肖磊. 疲劳裂纹尖端残余应力场的深度-传感压痕测试与有限元分析[J]. 航空学报, 2012, 33(12): 2211-2220.

LIU Jianzhong, YE Duyi, ZHANG Li' na, XIAO Lei. Depth-sensing Indentation Measurement and Finite Element Analysis of Residual Stress Field near Fatigue Crack Tip[J]. Acta Aeronautica et Astronautica Sinica, 2012, 33(12): 2211-2220.

Service

- ▶ 把本文推荐给朋友
- ▶ 加入我的书架
- ▶ 加入引用管理器
- ▶ Email Alert
- ▶ RSS

作者相关文章

- ▶ 刘建中
- ▶ 叶笃毅
- ▶ 张丽娜
- ▶ 肖磊

