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飞机结构疲劳试验载荷的优化设计

孟繁沛¹, 王建邦¹, 李令芳², 高建军²

1. 西安飞机工业集团 有限责任公司, 陕西西安 710089; 2. 西安飞机设计研究所, 陕西西安 710089

OPTIMUM DESIGN OF FATIGUE TESTING LOADS FOR AIRPLANE STRUCTURES

MENG Fan-pei¹, WANG Jian-bang¹, LI Ling-fang², GAO Jian-jun²

1. Xi'an Aircraft Industry Group Corporation, Xi'an 710089, China; 2. Xi'an Aircraft Design Institute, Xi'an 710089, China

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摘要 提出用对载荷误差求极值的方法实现最小误差控制的试验载荷计算,并通过精度加权系数综合控制误差的合理分配,完成试验载荷的优化设计。该方法已经应用于某型飞机疲劳试验的试验载荷计算,为该机全尺寸疲劳试验的成功奠定了基础。

关键词: 疲劳试验载荷 载荷误差 载荷精度加权系数 载荷优化方程组

Abstract: In full scale fatigue testing for airplane structures, the loads, which are used to simulate the load spectrum of various flying status in airplane service, are imposed on finite loading spots. Whether the loading is accurate or not, the testing loads calculation is needed to guarantee its precision. In this paper, the method of finding extremum for load error is proposed. The method realizes calculation of minimum error control. The load optimum design is completed through rational error distribution by using error synthetical control of weighted precision coefficients. This method is applied to fatigue testing loads calculation for a type of airplane. It is the basis of the successful full scale fatigue testing of the airplane.

Keywords: load of fat igue test ing load erro r w eigh ted coefficients of loading p recision op t imal load simultaneous equat ions

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