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多级考虑冷气掺混流片变厚度的 S_1 流面研究

Research on S_1 stream surface for cooling air mixing and variable thickness flow slice in multi-stage

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中文关键词: [\$S_1\$ 流面](#) [冷气掺混](#) [气动](#) [平面薄片](#) [涡轮](#) [优化](#)

英文关键词: [\$S_1\$ stream surface](#) [cooling air mixing](#) [aerodynamic](#) [plane slice](#) [turbine](#) [optimization](#)

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中文摘要:

为减少气冷涡轮气动设计难度,提出一套基于多级气冷涡轮考虑冷气掺混及随流道翘曲、变厚度的 S_1 流面计算思路,编制了带冷气的翘曲 S_1 流面薄片计算的参数化方法程序及网格自动生成程序,改良了传统平面薄片,对比分析了改良后平面薄片、翘曲 S_1 流面薄片以及三维计算间差异,对某高压涡轮进行了翘曲 S_1 流面薄片气动优化.结果显示:与三维计算对比,改良后平面薄片最大流量差距为22.68%,翘曲 S_1 流面薄片为3.58%,一维数据上翘曲 S_1 流面薄片更接近三维计算;型面压力分布及马赫数云图分布上翘曲面 S_1 流面薄片较改良后平面薄片更贴近三维计算;采用翘曲 S_1 流面薄片进行优化后,效率较原始方案提升0.41%,流量较原始方案仅增加0.21%.

英文摘要:

To reduce the difficulty of aerodynamic design in air-cooled turbine, a set of calculation ideas based on multi-stage air-cooled turbine were presented in consideration of the cooling air mixing, warped S_1 stream surface and variable thickness along with flow passage; the parameterization method program and automatic mesh generation program of warped S_1 stream surface slice with cooling condition were prepared, and the traditional plane slice was modified. Comparative analysis of the difference among the modified plane slice, warped S_1 stream surface slice and three-dimensional calculation was made, and aerodynamic design of a high-pressure turbine with warped S_1 stream surface slice was optimized. The results show that: in comparison with three-dimensional calculation, the maximum gap of mass flow in modified plane slice is up to 22.68%, and 3.58% in warped S_1 stream surface slice; the warped S_1 stream surface slice is closer to three-dimensional calculation on one-dimensional data; blade pressure and Mach number contour distribution of warped S_1 stream surface slice are closer to three-dimensional calculation than that of modified plane slice. After aerodynamic optimization in warped S_1 stream surface slice, compared with original model, the stage efficiency has been increased by 0.41%, while mass flow has only been increased by 0.21%.

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