



航空学报 » 2012, Vol. 33 » Issue (8) :1427-1433 DOI:

固体力学与飞行器总体设计

最新目录 | 下期目录 | 过刊浏览 | 高级检索

<< << 前一页 | 后一页 >> >>

复合材料机身壁板的纵向连接设计与失效分析

钱一彬¹, 钟小丹¹, 陈普会¹, 刘利阳², 王进²

1. 南京航空航天大学 航空宇航学院, 江苏 南京 210016;
2. 沈阳飞机设计研究所, 辽宁 沈阳 110035

Longitudinal Panel Splice Design and Failure Analysis of Composite Fuselage Structures

QIAN Yibin¹, ZHONG Xiaodan¹, CHEN Puhui¹, LIU Liyang², WANG Jin²

1. College of Aerospace Engineering, Nanjing University of Aeronautics & Astronautics, Nanjing 210016, China;
2. Shenyang Institute of Aircraft Design, Shenyang 110035, China

摘要

参考文献

相关文章

Download: [PDF \(1940KB\)](#) [HTML 1KB](#) Export: [BibTeX](#) or [EndNote \(RIS\)](#) [Supporting Info](#)

摘要 按照连接设计准则及机身压差载荷水平,开展了复合材料机身壁板的纵向连接设计研究。为提高壁板多钉连接结构分析精度及设计效率,发展了一种基于Fastener单元的钉群载荷计算方法,在此基础上结合单钉失效分析模型,提出了一种壁板多钉连接区的失效评估方法。首先,通过与试验数据对比,验证了采用Fastener单元求解钉群载荷的可行性;然后运用Fastener单元分析壁板连接结构的钉载分配;最后基于钉载分析结果,对局部危险区域采用单钉模型进行失效载荷计算并进而评估壁板连接区的失效载荷。本方法特别适用于快速、有效地校核多钉连接区的连接强度。

关键词: 复合材料 机身 壁板纵向连接 多钉连接 失效分析

Abstract: A study on the longitudinal panel splice design of composite fuselage structures is carried out in accordance with the design criteria of mechanical joints and pressure differential level. In order to improve the analysis precision and design efficiency for multiple-bolted joints of panel splice, a novel method for calculating the load distribution of multiple-bolted joints is developed by using fastener elements. Combined with failure analysis of single-bolted joints, an evaluation method is presented to predict the failure strength of panel splice joints. Firstly, the feasibility of using fastener elements to determine the load distribution of multiple-bolted joints is confirmed via comparing the FEM result with experimental data. Then the fastener load distribution of panel splice joints is studied by applying fastener elements. Finally, based on the previous analysis of fastener loads, the failure strength of panel splice joints is assessed by failure analysis on single-bolted joints at critical locations. The study provides a very simple and effective approach for strength calibration of multiple-bolted joints.

Keywords: composite fuselage longitudinal panel splice multiple-bolted joint failure analysis

Received 2011-10-13;

Fund:

国家自然科学基金(10872091)

Corresponding Authors: 陈普会 Email: phchen@nuaa.edu.cn

引用本文:

钱一彬, 钟小丹, 陈普会, 刘利阳, 王进. 复合材料机身壁板的纵向连接设计与失效分析[J]. 航空学报, 2012, 33(8): 1427-1433.

QIAN Yibin, ZHONG Xiaodan, CHEN Puhui, LIU Liyang, WANG Jin. Longitudinal Panel Splice Design and Failure Analysis of Composite Fuselage Structures[J]. Acta Aeronautica et Astronautica Sinica, 2012, 33(8): 1427-1433.

Service

- ▶ 把本文推荐给朋友
- ▶ 加入我的书架
- ▶ 加入引用管理器
- ▶ Email Alert
- ▶ RSS

作者相关文章

- ▶ 钱一彬
- ▶ 钟小丹
- ▶ 陈普会
- ▶ 刘利阳
- ▶ 王进

