

加入收藏丨设为首页

中国航空学会主办

首页 本刊介绍 编委会 投稿须知 审稿编辑流程 期刊征订 广告征订 English 选择皮肤: 🗋 🗖 🧧

Hide Expanded Menus

WU Xiang-yu,TAO Zhi,DU Zhi-neng,ZHANG Shu-lin,YOU Hong-de.Experiments of flow and heat transfer performances for air-cooling laminated turbine vane[J].航空动力学报,2014,29(5):1008~1013

Experiments of flow and heat transfer performances for air-cooling laminated turbine vane

Experiments of flow and heat transfer performances for air-cooling laminated turbine vane _{投稿时间: 2013-12-24}

DOI:

中文关键词: <u>lamilloy</u> <u>turbine vane</u> <u>flow performance</u> <u>heat transfer</u> <u>static cascade</u>

英文关键词:<u>lamilloy</u> <u>turbine vane</u> <u>flow performance</u> <u>heat transfer</u> <u>static cascade</u>

基金项目:

作者 单位

<u>WU Xiang-yu</u>	National Key Laboratory of Science and Technology on Aero-Engine Aero-thermodynamics, School of Energy and Power
	Engineering, <u>Beijing University of Aeronautics and Astronautics</u> , <u>Beijing 100191</u> , <u>China</u> ; <u>Shenyang Engine Design and</u>
	Research Institute, Aviation Industry Corporation of China, Shenyang 110015, China
<u>TAO Zhi</u>	National Key Laboratory of Science and Technology on Aero-Engine Aero-thermodynamics, School of Energy and Power
	Engineering, Beijing University of Aeronautics and Astronautics, Beijing 100191, China
<u>DU Zhi-neng</u>	Shenyang Engine Design and Research Institute, Aviation Industry Corporation of China, Shenyang 110015, China
ZHANG Shu-lin	Shenvang Engine Design and Research Institute, Aviation Industry Corporation of China, Shenvang 110015, China
<u>YOU Hong-de</u>	Shenyang Engine Design and Research Institute, Aviation Industry Corporation of China, Shenyang 110015, China

摘要点击次数: 44

全文下载次数: 56

中文摘要:

Flow resistance and heat transfer coefficients of typical double wall laminated film cooling configuration within a turbine vane w ere experimentally studied. The specimen was in large scale, and made of transparent organic glass. Laminated configuration consisted of double wall laminates, pin-fins, staggered arrays of impingement and film holes. The number ratio of impingement holes, pin-fins and fil m holes was 2:1:1. Five experiment vanes were installed in static cascade, and experiments were carried out under constant heat flux. Re of internal cooling air in the experiment was from 10^4 to 2×10^5 , and Re of external fluid was from 10^5 to 3×10^5 . The experiment results show that flow resistances of front channel and back channel of the vane are in the same level, and both of them decrease as Re of cooli ng air increases. Nu of front channel is slightly higher than that of back channel. Both of them increase as Re of cooling air increase s. And experiment results were obtained from experiment vanes were compared with that obtained from laminated flat plates, and the tende ncy of the results agrees well.

英文摘要:

Flow resistance and heat transfer coefficients of typical double wall laminated film cooling configuration within a turbine vane were experimentally studied. The specimen was in large scale, and made of transparent organic glass. Laminated configuration consisted of double wall laminates, pin-fins, staggered arrays of impingement and film holes. The number ratio of impingement holes, pin-fins and film holes was 21:1. Five experiment vanes were installed in static cascade, and experiments were carried out under constant heat flux. Re of internal cooling air in the experiment was from 10^4 to 2×10^5 , and Re of external fluid was from 10^5 to 3×10^5 . The experiment results show that flow resistances of front channel and back channel of the vane are in the same level, and both of them decrease as Re of cooling air increases. Nu of front channel is slightly higher than that of back channel. Both of them increase as Re of cooling air increases were obtained from experiment vanes were compared with that obtained from laminated flat plates, and the tendency of the results agrees well.

查看全文 查看/发表评论 下载PDF阅读器

关闭

