

DPIV技术在高超声速通气模型内阻测量中的应用(PDF下载)

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Title: Applications of DPIV technique in internal drag measurement for hypersonic flowthrough model

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摘要: 针对高超声速通气模型内阻测量存在的误差大的问题,首次采用DPIV技术和总压测量排架相结合的方法测量了超燃发动机尾喷管流场速度矢量和皮托压力分布,间接获得了内流出口处平均马赫数和静压平均值,从而实现了高超声速通气模型内阻测量。研究结果表明:DPIV试验获得的粒子图像可以清楚地显示喷管出口位置内外流的分界面、内外流混合层的尾迹、通气模型外表面边界层;DPIV试验获得的速度矢量场结果准确、精度高,能够提供远远超出传统测量技术所能提供的流场信息。DPIV技术作为一种有力的测量手段,在高超声速飞行器研究中能够发挥重大的作用。

参考文献/REFERENCES

- [1] Small W J, Weidner J P, Johnston P J.Scramjet nozzle design and analysis as applied to a highly integrated hypersonic research airplane[NASA TN D-8334[R].Washington DC: NASA, 1976.
- [2] Holland S D, Woods W C, Engelund W C.Hyper-X research vehicle experimental aerodynamics test program overview[J].Journal of Spacecraft and Rockets, 2001, 38(6): 11-12.
- [3] Engelund W C, Holland S D, Cockrell C E, et al.Aerodynamic database development for the Hyper-X airframe integrated scramjet propulsive experiments[C]//Proceedings of the 18th AIAA Applied Aerodynamics Conference.Denver, Colorado: AIAA, 2000: AIAA 2000-4006.
- [4] 范洁川·风洞试验手册[M].北京: 航空工业出版社, 2002.(Fan Jiechuan.Handbook of wind tunnel test[M].Beijing: Aviation Industry Press, 2002.(in Chinese)).
- [5] 程厚梅·风洞实验干扰与修正[M].北京: 国防工业出版社, 2003.(Cheng Houmei.Interference and correction on wind tunnel testing[M].Beijing: National Defence Industry Press, 2003(in Chinese)).
- [6] Adrian R J.Particle image techniques for experimental fluid mechanics[J].Annual Review of Fluid Mechanics, 1991, 23: 261-304.
- [7] 田立丰, 易仕和, 赵玉新, 等, 超声速光学头罩流场的PIV研究[J].实验流体力学, 2010, 24(1): 26-29.(Tian Lifeng, Yi Shihe, Zhao Yuxin, et al.PIV study of supersonic flow around an optical bow cap[J].Journal of Experiments in Fluid Mechanics, 2010, 24(1): 26-29 (in Chinese)).
- [8] Sahoo D, Desai P, Smits A.Experimental investigation of helium injection in a hypersonic turbulent boundary layer[C]//Proceedings of the 48th AIAA Aerospace Sciences Meeting including the New Horizons Forum and Aerospace Exposition.Orlando, Florida: AIAA, 2010: AIAA 2010-1559.
- [9] Van Der Draai R K, Van Schinkel R, Telesca A, et al.Application of PIV in (local) supersonic flows in DNW wind tunnels[C]// Proceedings of the 41st AIAA/ASME/SAE/ASEE Joint Propulsion Conference and Exhibit.Tucson, Arizona: AIAA, 2005: AIAA 2005-4082.
- [10] Haertig J, Havermann M, Ray C, et al.PIV measurements in Mach 3.5 and 4.5 shock tunnel flow[C]//Proceedings of the 39th Aerospace Sciences Meeting and Exhibit.Reno, Nevada: AIAA, 2001: AIAA 2001-0699.

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